

# Numerical analysis of a DFB fiber laser sensor

Sorin Miclos, Dan Savastru, and Ion Lancranjan

**Abstract**—This paper is pointing to numerical simulation of various aspects of distributed feedback fiber laser sensors and their applications mainly in the field of the aeronautical applications. The developed numerical analysis has the aim of a better understanding of DFB-FL itself and of its interaction with environment in order to be operated as a sensor. Numerical analysis concentrates both on the FEM and phenomenological methods.

**Keywords**— FBG, DFB-FL, DBR-FL, sensor, COMSOL.

## I. INTRODUCTION

**D**ISTRIBUTED feedback fiber lasers (DFB-FL) and distributed Bragg reflector fiber lasers (DBR-FL) possess certain unique properties that make them quite attractive for a number of different applications. They are inherently fiber compatible, and very simple passive thermal stabilization is sufficient to ensure the stability of the laser [4]-[10].

A number of different active dopants such as erbium, ytterbium, neodymium and thulium can be used in order to cover different windows of the optical spectrum. These features, combined with the ability to define the emitted wavelength precisely through the grating structure along with the narrow linewidth and low relative intensity noise (RIN), make DFB-FL and DBR-FL very advantageous for telecommunication and sensor applications [1]-[3]. In addition, a number of DFB fiber lasers can be configured in a parallel array to provide flexibility in pumping conditions and provide pump redundancy [2], [4].

Robust single polarization and narrow linewidth of DFB lasers are very desirable for sensor systems [5]-[7]. Alternatively, DFB lasers can be made to operate in stable dual polarization regime so that simultaneous measurements can be carried out [8]-[10],[26]. In addition to the sensing and telecom applications, DFB fiber lasers suitable for high-power applications have been demonstrated [11].

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## II. SPECIFICATION OF DFB-FL AND DBR-FL SENSOR APPLICATION

An important aeronautical application of fiber optic sensors consists in determination of *transition zone* between *laminar* and *turbulent flow* of air along the wing surface. Intermittent regime occurring in-between these two regions (transition) is characterized by turbulent bursts in laminar flow.

The basic idea of this type of measurement is to evaluate the pressure variation in the two zones:

1. Laminar flow - relative constant value of air static pressure, low frequency ( $\sim 100$  Hz) and small amplitude ( $\Delta P \sim 1$  Pa) pressure variations.
2. Turbulent flow - larger and nonstationary value of air static pressure, higher frequency ( $\sim 10$  kHz) and higher amplitude ( $\Delta P \sim 10$  Pa) pressure variations.

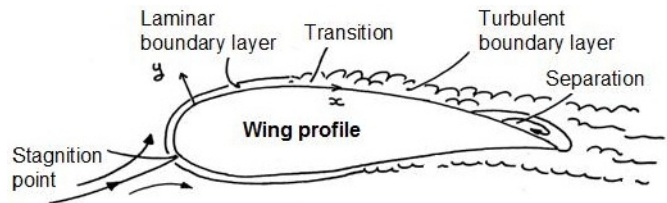


Fig. 1 Schematic representation of the main investigated aeronautical application of DFB-FL and DBR-FL

The main investigated aeronautical DFB-FL and DBR-FL sensors application consists in determination of the transition zone (line) between laminar and turbulent air flow along the aircraft wing surface. The laminar and turbulent boundary layers can be observed in Fig. 1.

Possible fiber optic “reaction”: linear glass strain deformation (glass Young’s modulus of elasticity is  $E = 50 \div 90 \times 10^9$  N/m<sup>2</sup>) under air turbulent pressure bursts (deformations of  $10^{-9} \div 10^{-8}$  m) is extremely difficult to measure even by optical interferometer methods. In this situation micro-bending of fiber optic appears to be more feasible deformation as an effect of turbulent air flow pressure bumps. Schematic representation of the main investigated aeronautical application of DFB-FL and DBR-FL is presented in Fig. 2. The laminar and turbulent air flow zones along the aircraft wing surface are indicated. One possible position of the fiber optic sensor can be observed. Analyzing the layout presented in Fig. 2 some rough estimations of the force applied to the FBG sensor can be made. For a 10 mm length of the single mode fiber optic sen-

For a sensor segment a force of  $1.25 \mu\text{N}$  for the laminar air flow and  $12.5 \mu\text{N}$  for the turbulent air flow can be estimated.

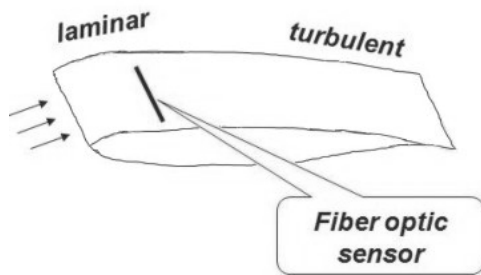


Fig. 2 Schematic representation with few relevant insights of the main investigated aeronautical application of DFB-FL and DBR-FL

In Fig. 3 it can be observed that the fiber optic sensor is embedded close ( $0.2 \text{ mm}$  depth) to the wing surface. The fiber optic sensor is placed into a soft material, like paraffin, under an  $0.2 \text{ mm}$  thick aluminum foil.

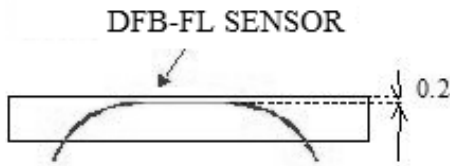


Fig. 3 Insights of one possible way of mounting the DFB-FL and/or DBR-FL in the wing for determination of transition zone between laminar and turbulent air flow along the aircraft wing surface

One possible procedure for reading the fiber optic sensor is presented schematically in Fig. 4. This possible procedure is based on precise evaluation of lasing wavelength, lasing, which depends on the laser resonant cavity length.

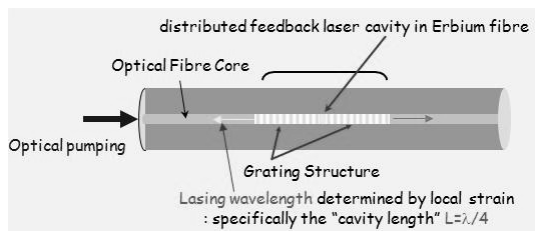


Fig. 4 Some insights about the structure of the DFB-FL and/or DBR-FL proposed to be used for the determination of the transition zone between laminar and turbulent air flow along the wing surface

Some additional insights about the structure of the DFB-FL and/or DBR-FL proposed to be used for the determination of the transition zone between laminar and turbulent air flow along the wing surface are displayed in Fig. 5. The possible fiber optic sensor output reading by measuring the lasing wavelength shift ( $\Delta\lambda$ ) is indicated.

It is to be noted the role of pumping wavelength used for DFB-FL or DBR-FL. This output reading is applicable for both diode pumping wavelengths, namely  $980 \text{ nm}$  or  $1480 \text{ nm}$

The first one,  $980 \text{ nm}$  wavelength, is more efficient than the second but has lower saturation intensity. The second one,  $1480 \text{ nm}$  wavelength, seems to be more interesting for sensor application because its more extended linearity response domain.

An important observation is that the pressure bumps of the turbulent air flow can be recorded by DFB-FL or DBR-FL in two possible ways:

- in the Bragg grating zone;
- in the zone between two successive such Bragg gratings

### III. DFB-FL AND DBR-FL SENSOR ARCHITECTURE

Regarding the Distributed Feedback Fiber Laser (DFB-FL) and Distributed Bragg Reflector Fiber Laser (DBR-FL) sensors architecture the following are to be observed:

- Both are built using single-mode optical fiber (core of  $5 - 10 \mu\text{m}$  diameter and cladding of  $80 - 100 \mu\text{m}$  overall diameter)
- Both are built using single-mode optical fiber as active medium. The active medium is formed by doping the core of the optical fiber with erbium ions ( $\text{Er}^{3+}$ ).

The important feature consists in the Bragg grating – spatial sinusoidal refractive index variation in and along the core of the optical fiber. Bragg grating characteristic parameters are:  $\Lambda$  – the wavelength of spatial modulation of the refractive index,  $\lambda_B$  – the Bragg wavelength (defined as  $2 \cdot n_{\text{eff}} \cdot \Lambda$ , the wavelength of maximum reflection coefficient),  $n_{\text{eff}}$  – the effective value of the refractive index, corresponding to the fundamental mode of electromagnetic field propagation into the optical fiber, being imposed by the geometric characteristics of the optic fiber).

DBR-FL means a laser oscillator formed by the optical fiber active medium placed between two mirrors Bragg gratings (distributed reflector) while DFB-FL means a laser oscillator formed by the optical fiber active medium support of the Bragg grating [5]-[10],[13],[14].

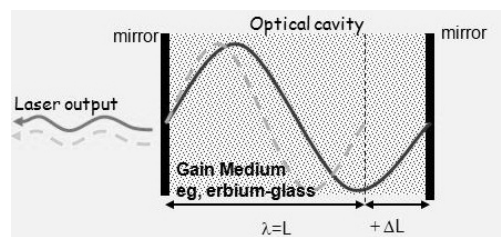


Fig. 5 The possible fiber optic sensor output reading by measuring the lasing wavelength shift ( $\Delta\lambda$ )

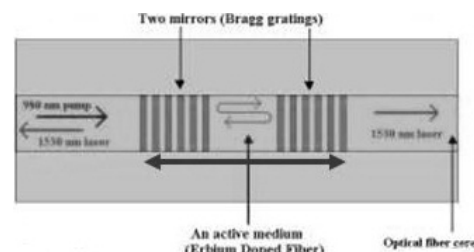


Fig. 6 The possible fiber optic sensor output reading by measuring the lasing wavelength shift ( $\Delta\lambda$ )

IV. DFB-FL THEORY

Traditionally, there have been three main DFB laser cavity designs that offer different performance and distinctive operational characteristics, presented in the followings.

It was recently shown that the classic parametric optimization approach for a DFB laser, i.e., the definition of the optimum resonator geometry and dimensional values, is analogous to Rigrod optimization [17],[18] of reflectivities in Fabry-Pérot laser cavities of fixed length. It can also be shown that it is possible to further improve the DFB laser efficiency by increasing the effective cavity length without changing the total device length and optimum reflectivities, using a step-apodized profile.

Both optimization approaches are parametric in nature. The main cavity features are defined a priori, and their parameters are continuously varied until a maximum efficiency is reached. However, none of these approaches guarantees that the ultimate, i.e., maximum possible, efficiency for the given medium has been achieved. In this paper, a drastically different approach is followed.

The new method follows an “inverse scattering” philosophy in that, for a given medium and pumping arrangement, it first derives the maximum possible efficiency and the use of the developed algorithm defines the required generalized DFB cavity. This is achieved without any significant a priori assumptions about the grating characteristics. Taking into account the local pump power, the method relies on the calculation of the optimum intracavity signal distribution that results at maximum pump-to-signal conversion at every point along the cavity.

Using this information, the developed algorithm calculates the required grating strength distribution that results in the desired optimum signal, pump, and gain distribution.

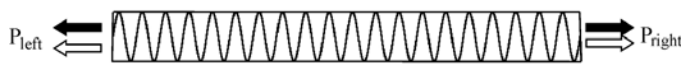


Fig. 7 Refractive index profile for conventional DFB laser designs. The classic design and two-wavelength bidirectional operation

The classic design and two-wavelength bidirectional operation is displayed in Fig. 7. It consists of a uniform refractive index grating, with constant amplitude and constant period, incorporated in an active medium. This type of DFB laser operates at two fundamental longitudinal modes at different wavelengths, corresponding to the edges of the grating band-gap, and gives symmetric output powers from both ends, which are equally divided between these two modes [12]. Such a cavity provides dual-wavelength bidirectional operation.



Fig. 8 Refractive index profile for conventional DFB laser designs. Symmetric-phase shifted design and single-wavelength bidirectional operation

Fig. 8 shows the symmetric-phase shifted design and single-wavelength bidirectional operation. In practice, however, single-wavelength operation is desirable. This is achieved by introducing a  $\pi$ -shift in the spatial phase of the grating [13]–[15]. If the phase shift is located in the middle of the grating due to the symmetry of the cavity, the output powers at both ends are equal. Such a cavity provides single-wavelength operation, coinciding with the grating Bragg wavelength, and bidirectional operation.

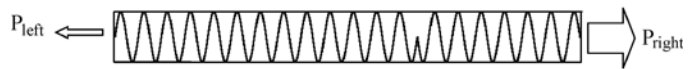


Fig. 9 Refractive index profile for conventional DFB laser designs. Asymmetric  $\pi$ -phase-shifted design and single-wavelength unidirectional operation

Asymmetric  $\pi$ -phase-shifted design and single-wavelength unidirectional operation is shown in Fig. 9. In addition to single-wavelength emission, unidirectionality is a very desirable feature of high-performance lasers. By placing the phase shift asymmetrically with respect to the grating center, as shown in Fig. 9, larger output power is obtained from the shorter end [10], [16]. In this asymmetric design, the maximum output power from the desired end is obtained for a particular phase-shift position and coupling coefficient value. Optimum values of parameters and are found by varying them over a defined range, either by simulation or by experiment.

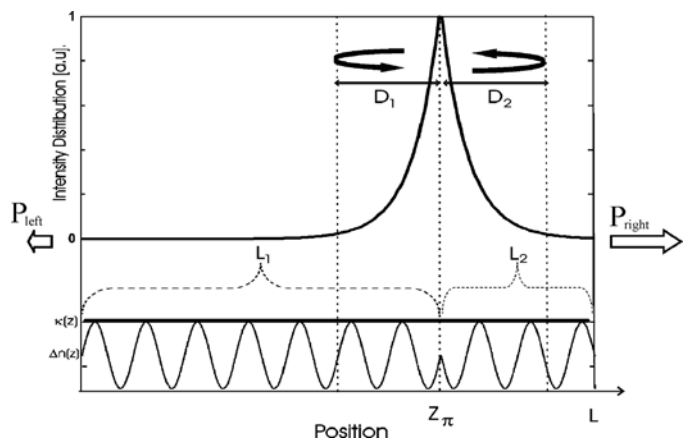


Fig. 10 Standard asymmetric DFB-FL structure

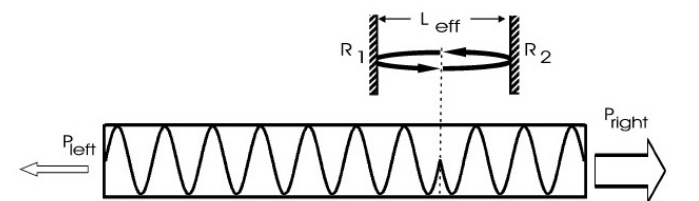


Fig. 11 Standard asymmetric DFB-FL structure.  $L_{eff}$  represents the sum of the electromagnetic field penetration depth into Bragg gratings

Standard asymmetric DFB-FL structure is illustrated in Fig. 10 and 11. The optimum position of the  $\pi$ -phase shift position ( $z_p$ ) can be observed.  $D_1$  and  $D_2$  represent the “penetration” depth of electromagnetic field into the Bragg grating zones.

In Fig. 12 is presented Apodized standard asymmetric DFB-FL structure.  $L_{\text{eff}}$  represents the sum of the electromagnetic field penetration depth into Bragg gratings. Apodization consists in modification of refractive index spatial modulation depth (amplitude).

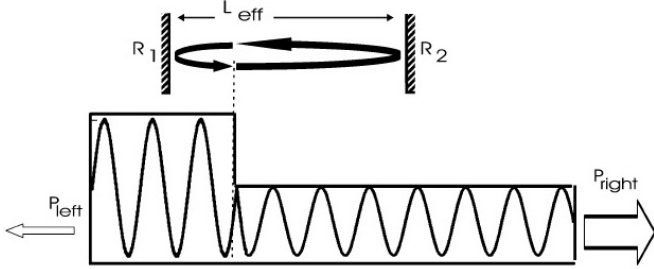


Fig. 12 Apodized standard asymmetric DFB-FL structure

The standard coupled-mode equations for counter-propagating fields are used (see, e.g., [20]). The electric field ( $E$ ) is the sum of two counterpropagating fields ( $A$  and  $B$ ) [13]-[25].

The forward-propagating field amplitude equation of propagation is given by equation:

$$\frac{dA(z)}{dz} = \alpha(z)A(z) + \kappa(z)B(z)e^{i\Gamma(z)} \quad (1)$$

The backward-propagating field amplitude equation of propagation is:

$$\frac{dB(z)}{dz} = -\alpha(z)B(z) + \kappa(z)A(z)e^{-i\Gamma(z)} \quad (2)$$

where  $A(z)$  is the amplitude of the forward-propagating field,  $B(z)$  is the amplitude of the backward-propagating field,  $A(z)e^{-i\Gamma(z)}$  represents the envelope of the forward-propagating field,  $B(z)e^{i\Gamma(z)}$  represents the envelope of the backward-propagating field,  $\alpha(z)$  is the field gain,  $\kappa(z)$  is the coupling coefficient while  $\Gamma(z)$  is the spatial phase factor or coefficient. A schematic representation of coupled-mode procedure or method, used for numerical evaluation of DFB-FL structure is presented in Fig. 13.

Designating by  $\alpha(z)$  the net field gain including the background loss and  $\phi(z)$  the Bragg grating phase, the spatial phase factor/coefficient  $\Gamma(z)$  will be given by this equation, where  $\beta$  is the unperturbed waveguide mode:

$$\Gamma(z) = 2\beta(z) - \phi(z) \quad (3)$$

The equation defining the Bragg grating phase  $\phi(z)$  is:

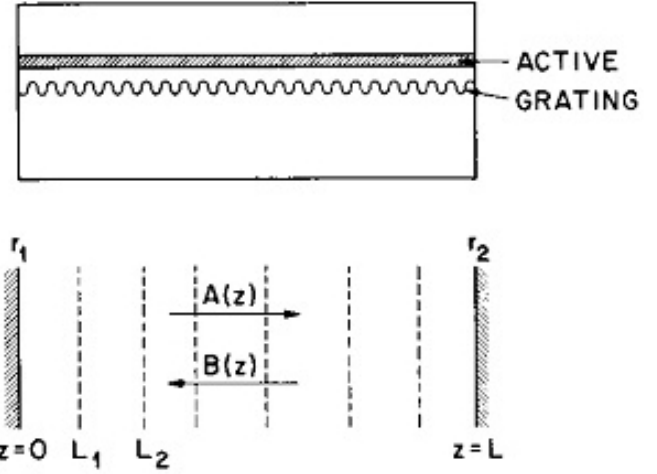


Fig. 13 Schematic representation of coupled-mode procedure/method

$$\phi(z) = \int_0^z \frac{2\pi}{\Lambda(z^*)} dz^* \quad (4)$$

where  $\Lambda(z)$  represents the local grating period. The average signal intensity definition is:

$$S(z) = A^2(z) + B^2(z) \quad (5)$$

while the definition of the intensity difference between the counterpropagating fields is:

$$D(z) = A^2(z) - B^2(z) \quad (6)$$

The intensity difference  $D(z)$  can be expressed as:

$$D(z) = D(0) + 2 \int_0^z \alpha(z^*) \cdot S(z^*) dz^* \quad (7)$$

The standard coupled-mode propagation equations for counterpropagating fields are can be manipulated to provide expressions for  $k(z)$ , the coupling coefficient of the electromagnetic field:

$$k(z) = \frac{\frac{dS(z)}{dz} - D(z)\alpha(z)}{\cos(\Gamma(z))\sqrt{S^2(z) - D^2(z)}} \quad (8)$$

The usual DFB laser boundary conditions are:

$$A(0) = B(L) = 0 \quad (9)$$

The new/transformed DFB laser boundary conditions are:

$$\begin{aligned} D(0) &= -B^2(0) = 0 \\ D(L) &= A^2(L) = S(L) \end{aligned} \quad (10)$$

These boundary conditions represent the basis of our design method. Given  $S(z)$ ,  $\alpha(z)$  and  $A(z)$ , we can use them to find  $D(z)$  and then the required coupling coefficient distribution can be calculated:

$$n(z) = n_0 + \Delta n(z) \cdot \cos(\phi(z)) \quad (11)$$

The coupling coefficient defines the amount of the periodic perturbation required. If this perturbation is sinusoidal the varying refractive-index modulation in the form is defined by the above equation.  $n_0$  is the effective refractive index and  $\Delta n$  is the modulation amplitude [13], [15]-[17].

The reflection coefficient of a grating with constant gain at the Bragg wavelength is:

$$r = \frac{-k \cdot \sinh(\gamma L)}{\gamma \cdot \cosh(\gamma L) - \alpha \cdot \sinh(\gamma L)} \quad (12)$$

Here  $\gamma$  coefficient is  $\gamma = \sqrt{k^2 + \alpha^2}$ .

The approximation of reflection coefficient of a grating with constant gain at the Bragg wavelength is given by  $r \approx -\tanh(kL)$ .

The necessary condition for the validation of the above equation is  $\alpha \ll \kappa$ .

The reflectivity of the Bragg grating is equal to the reflectivity of a passive grating with no gain:

$$R = r^2 \approx \tanh^2(kL) \quad (13)$$

Due to the distributed nature of the reflection process in gratings, the incident wave penetrates into the grating before reemerging at the front end. It refers to the case of the case of constant gain and at the Bragg wavelength:

$$D = \frac{1}{2} \cdot \frac{\alpha L \left( \frac{\tanh(\gamma L)}{\gamma L} - \frac{1}{\cosh^2(\gamma L)} \right) + \tanh^2(\gamma L)}{\alpha \cdot \tanh^2(\gamma L) + \gamma \cdot \tanh(\gamma L)} \quad (14)$$

$$D \approx \frac{\tanh(kL)}{2k} = \frac{r}{2k} \quad (15)$$

In the case of a phase-shifted DFB laser, the total length of effective cavity in which the fields are circulating is:

$$L_{eff} = D_1 + D_2 \approx \left( \frac{|r_1|}{2k_1} + \frac{r_2}{2k_2} \right) \quad (16)$$

$D_1$  and  $D_2$  are the penetration depths into the Bragg grating segments on the left-hand side and on the right-hand side of the phase shift, respectively.

In the case of a uniform refractive index profile, the coupling coefficient is constant.

By observing the fact that the electromagnetic field is propagating through surfaces of  $\sim 10^{-12} \text{ m}^2$  area, nonlinear effects produced mainly into the core glass have to be considered. The most encountered such effect is birefringence [21]-[22], [25]. This effect can divide the peak of the Bragg grating reflection spectrum into two parts at the Bragg wavelength. Since the Bragg wavelength depends on the grating period and the effective index, physical and environmental effects such as temperature, stress and stretching the Bragg wavelength can change as:

$$d\lambda_B = \Delta\lambda_p dP + \Delta\lambda_T dT \quad (17)$$

where  $\Delta\lambda_p$  and  $\Delta\lambda_T$  represent the pressure and temperature variation coefficients of Bragg wavelength.  $\Delta\lambda_p$  and  $\Delta\lambda_T$  are defined as

$$\Delta\lambda_p = 2\lambda_{B,0} \left( \frac{\partial n_{eff}}{\partial P} \right) + 2n_{eff,0} \left( \frac{\partial \lambda_B}{\partial P} \right) \quad (18)$$

$$\Delta\lambda_T = 2\lambda_{B,0} \left( \frac{\partial n_{eff}}{\partial T} \right) + 2n_{eff,0} \left( \frac{\partial \lambda_B}{\partial T} \right) \quad (19)$$

In the above equations,  $P$  and  $T$  denote pressure and temperature respectively. Birefringence is due to the changes in propagation constants of the guided light. The value of birefringence ( $B$ ) along the  $z$ -axis is defined as:

$$B = B_0 + \frac{|\Delta n_{\parallel} + \Delta n_{\perp}|}{n_0} \quad (20)$$

where  $B_0$  is the unperturbed birefringence value and  $\Delta n_{\perp}$  and  $\Delta n_{\parallel}$  are the parallel and perpendicular refraction indices. Finally,  $\Delta n_{\perp}$  and  $\Delta n_{\parallel}$  differences will induce Bragg wavelength variation.

An important detail of the performed numerical and theoretical analysis has to be mentioned. It is concerning at field gain coefficient  $\alpha(z)$ . It has to be calculated by considering the phenomena which are produced during the erbium ions pump cycle, namely the up-conversion from second and third  $\text{Er}^{3+}$  excited levels and, the second kind of them, the excited state absorption of pumping radiation. In Table I coefficients describing these ionic scale processes are defined.

## V. DBR-FL REFLECTOR THEORY

A mode propagating on a straight fiber or waveguide fabricated from non-absorbing, non-scattering materials will in principle propagate indefinitely without any loss of power. However, if a bend is introduced, the translational invariance is broken and power is lost from the mode as it propagates into, along and out of the bend. This applies to the fundamental mode in the case of single-mode fibers and waveguides and to all bound modes in the case of bent multimode fibers or

waveguides [21].

Two types of optic fiber bend losses can be considered [20], [12] – [16]:

- Transition loss is associated with the abrupt or rapid change in curvature at the beginning and the end of a bend;
- Pure bend loss is associated with the loss from the bend of constant curvature in between the optic fiber.

The transition loss can be described by an abrupt change in the curvature  $k$  from the straight waveguide ( $k \sim 0$ ) to that of the bent waveguide of constant radius  $R_b$  ( $k = 1/R_b$ ). The fundamental-mode field is shifted slightly outwards in the plane of the bend, thereby causing a miss-match with the field of the straight waveguide, as presented in Fig. 14.

The fractional loss in fundamental-mode power,  $\delta P/P$ , can be calculated from the overlap integral between the fields. Within the Gaussian approximation to the fundamental mode field and assuming that the spot size  $s$  and core radius or half-width  $\rho$  are approximately equal, where  $V$  is the fiber or waveguide parameter and  $D$  is the relative index difference this gives:

$$\frac{\delta P}{P} \approx \frac{1}{16} \cdot \frac{V^4}{\Delta^2} \cdot \frac{\rho^2}{R_b^2} \quad (21)$$

Minimizing transition loss can be achieved by considering a number of techniques for significantly reducing transition loss. In the case of planar waveguides it is often possible to fabricate the bend so that there is an abrupt offset between the cores of the straight and bent waveguides in the plane of the bend. In Fig. 14 this can be seen as being equivalent to displacing the bent core downwards so that the two fundamental-mode fields overlap. Alternatively, if a gradual increase in curvature is introduced between the straight and uniformly bent sections, the fundamental field of the straight waveguide will evolve approximately adiabatically into the offset field of the uniformly bent section.

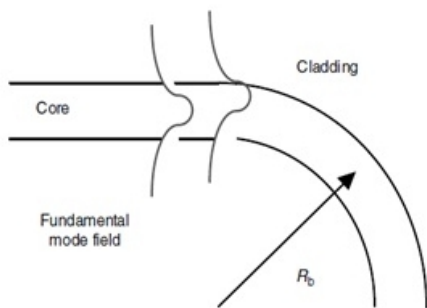


Fig. 14 Outward shift in the fundamental-mode electric field on entering a bend

The pure bend loss is defined by the fundamental mode continuously optical power loses when propagating along the curved path of the core of constant radius  $R_b$ . It is assumed that the cladding is essentially unbounded and not affected by the fiber optic bend, keeping a constant cladding refractive index value,  $n_{cl}$ . The radiation loss increases rapidly with de-

creasing bend radius and occurs predominantly in the plane of the bend; in any other plane the effective bend radius is larger and hence the loss is very much reduced, as presented in Fig. 15.

It has to be observed that the phase velocity anywhere on the modal phase front rotating around the bend cannot exceed the speed of light in the cladding. Hence, beyond radius  $R_{rad}$  the modal field must necessarily radiate into the cladding, the radiation being emitted tangentially.

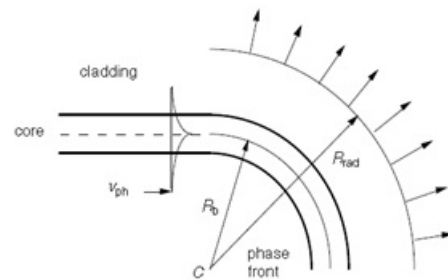


Fig. 15 Schematic of the bending effect of a fiber laser

The interface between the guided portion of the modal field around the bend and the radiated portion at  $R_{rad}$  is known as the radiation caustic, and it is the apparent origin of radiation. Between the core and the radiation caustic, the modal field is evanescent and decreases approximately exponentially with increasing radial distance from  $C$ . As the bend radius increases, the radiation caustic moves farther into the cladding, and the level of radiated power decreases.  $R_{rad}$  can be defined by the equation:

$$R_{rad} = \frac{C}{\Omega \cdot n_{cl}} \quad (22)$$

The present theoretical analysis is developed by considering step-index optical fibers (with a step profile of the refractive index). In terms of the core and cladding modal parameters  $U$  and  $W$ , respectively, relative index difference  $\Delta$ , core radius  $\rho$ , fiber parameter  $V$  and the bending radius  $R_b$ , an approximate expression for  $\gamma$  for the fundamental mode of a step-index fiber has the form [20 – 22]:

$$\gamma = \sqrt{\frac{\pi\rho}{R_b} \frac{V^2 \sqrt{W}}{2\rho U^2}} \exp\left(-\frac{4}{3} \Delta \frac{R_b W^3}{\rho V^2}\right) \quad (23)$$

where  $R_b$  is necessarily large compared to  $\rho$  because it is not possible to bend a fiber into a radius much below 10 mm without breakage. The pure bend loss coefficient is most sensitive to the expression inside the exponent because  $R_b$  and  $\rho$ . Loss decreases very rapidly with increasing values of  $R_b$  or  $\Delta$  or  $V$  (since  $W$  also increases with  $V$ ), and becomes arbitrarily small as  $R_b$  tends to infinity.

## VI. NUMERICAL SIMULATIONS RESULTS

The parameters of the investigated active FBG sensor are presented in Table I.

TABLE I  
ACTIVE FBG MODEL PARAMETER

Parameter	Value
Pump wavelength	$0.978 \dots 0.980 \cdot 10^{-6}$ m
Signal wavelength	$1.551 \dots 1.554 \cdot 10^{-6}$ m
Absorption cross-section	$3 \cdot 10^{-25}$ m <sup>2</sup>
Er <sup>3+</sup> concentration	$3 \cdot 10^{-25}$ m <sup>3</sup>
Er <sup>2</sup> I <sub>13/2</sub> life-time	10 <sup>-2</sup> s
Er <sup>2</sup> I <sub>11/2</sub> life-time	$0.6 \cdot 10^{-6}$ s
Er <sup>2</sup> F <sub>7/2</sub> radiative transition rate	$1.4 \cdot 10^6$ s <sup>-1</sup>
Er <sup>2</sup> F <sub>7/2</sub> non-radiative transition rate	10 <sup>3</sup> s <sup>-1</sup>
Er <sup>4</sup> S <sub>3/2</sub> radiative transition rate	$1.4 \cdot 10^6$ s <sup>-1</sup>
Er <sup>4</sup> S <sub>3/2</sub> non-radiative transition rate	10 <sup>3</sup> s <sup>-1</sup>
Er <sup>2</sup> H <sub>11/2</sub> radiative transition rate	$1.4 \cdot 10^6$ s <sup>-1</sup>
Er <sup>2</sup> H <sub>11/2</sub> non-radiative transition rate	10 <sup>3</sup> s <sup>-1</sup>
Up-conversion level 2 coefficient	10 <sup>-24</sup> m <sup>3</sup> s <sup>-1</sup>
Up-conversion level 3 coefficient	10 <sup>-24</sup> m <sup>3</sup> s <sup>-1</sup>
Excited state absorption level 2 coefficient	10 <sup>-27</sup> m <sup>3</sup> s <sup>-1</sup>
Excited state absorption level 3 coefficient	10 <sup>-27</sup> m <sup>3</sup> s <sup>-1</sup>
Signal background loss	0.15 m <sup>-1</sup>
Core radius	$2.3 \cdot 10^{-6}$ m
Core refractive index	1.4509
Cladding refractive index	1.4378
Refractive index spatial modulation wavelength	$1.5 \cdot 10^{-6}$ m

In Figure 16 the Bragg wavelength variations due to applied force of different strength are presented. It is to be observed that a Bragg wavelength variation of ~0.6 nm can be obtained for an applied force of 100 N.

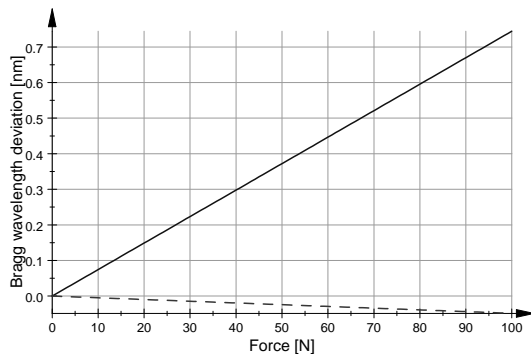


Fig. 16 The Bragg wavelength variations versus applied force. Solid line represents  $\Delta n_{\perp}$  while the dashed line represents  $\Delta n_{\parallel}$

Two types of an active FBG sensor numerical analysis were performed. The first one is focused on using the mathematical apparatus defined in Section IV pointing to evaluation of active FBG spectral characteristics versus environment parameters (pressure and temperature). The second type relies on evaluating the laser intensity field distribution inside the optic fiber with the purpose of evaluation of laser power.

For second type of numerical analyses two numerical simulation procedures were used:

- one relying on MATLAB - MuPAD software package, based on the above mentioned equations;
- the second one relying on COMSOL software packages.

Numerical simulations were performed for optical fiber with and without doping with erbium ions (Er<sup>3+</sup>). No significant differences were observed for doped or undoped optical fibers. The numerical simulations were performed using 1.550  $\mu$ m as the laser wavelength.

In the first stage, transition loss was simulated. Using (16) relative input power variation was calculated as:

$$P_{rel} = \frac{V^4}{16 \cdot \Delta^2} \cdot \frac{\rho^2}{R_b^2} \quad (24)$$

where  $\rho = 5$   $\mu$ m is the core radius,  $R_b = 5$  mm is the radius of curvature, while  $\Delta$  – relative index difference and  $V$  – modal parameter are calculated as it follows:

$$\Delta = \frac{n_{core}^2 - n_{clad}^2}{n_{core}^2} \quad (25)$$

$$V = \frac{2\pi \cdot \rho}{\lambda} \sqrt{n_{core}^2 - n_{clad}^2} \quad (26)$$

$n_{core} = 1.4457$  is the refractive index of the core, with a diameter of 10  $\mu$ m,  $n_{clad} = 1.4378$  is the refractive index of the cladding with an external diameter of 80  $\mu$ m, while  $\lambda = 1.55$   $\mu$ m denotes the wavelength. Fig. 17 illustrates the variation of relative input power  $P_{rel}$  vs. radius of curvature  $R_b$ .

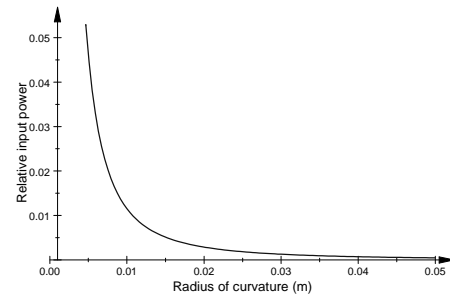


Fig. 17 Relative input power vs. radius of curvature

The numerical simulation performed using COMSOL Multiphysics is aiming to obtain an insight on the laser intensity distribution across the transverse section of the optic fiber. The option **2D** was used for the **Space Dimension**. Then the **RF Module** -> **Perpendicular Waves** -> **Hybrid-Mode Waves** -> **Mode analysis** options was used. The geometry of the transverse optical fiber cross section was developed considering realistic parameters. Elliptical deformation of the optical fiber was considered in order to resemble the bend.

Only numerical simulation of single mode optical fiber was considered. The developed geometry of the studied optical fiber was as realistic as possible. Nevertheless only axis symmetric optic fiber was considered. This means that, at this stage of development of DFB-FL and DBR-FL numerical simulation the point-by-point description of transverse fiber optic profile was neglected. In the future stage of development this more realistic geometry will be considered.

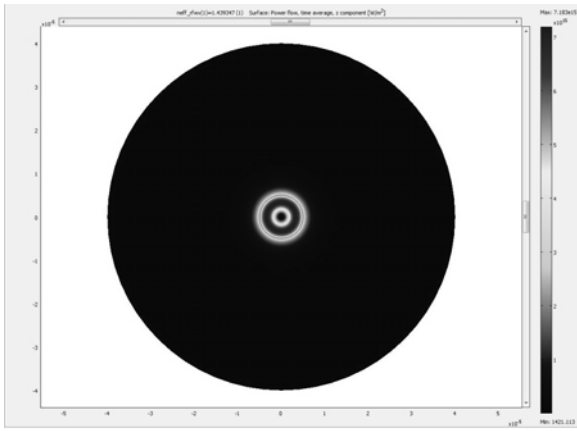


Fig. 18 The numerical simulated time averaged laser power flow across the transverse section of a singlemode optical fiber with a core of  $10\ \mu\text{m}$  diameter and a cladding of an overall  $80\ \mu\text{m}$  diameter

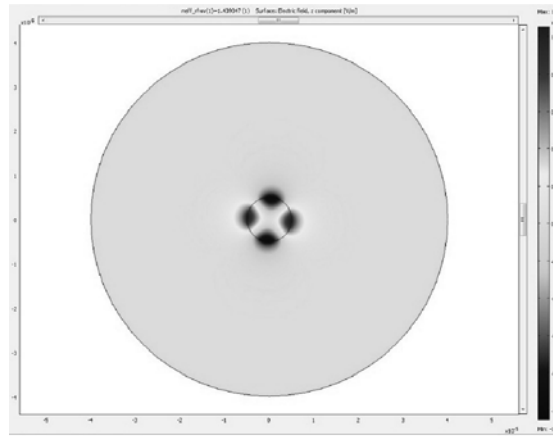


Fig. 19 The numerical simulated time averaged laser electric field distribution into the transverse section of a singlemode optical fiber with a core of  $10\ \mu\text{m}$  diameter and a cladding of an overall  $80\ \mu\text{m}$  diameter

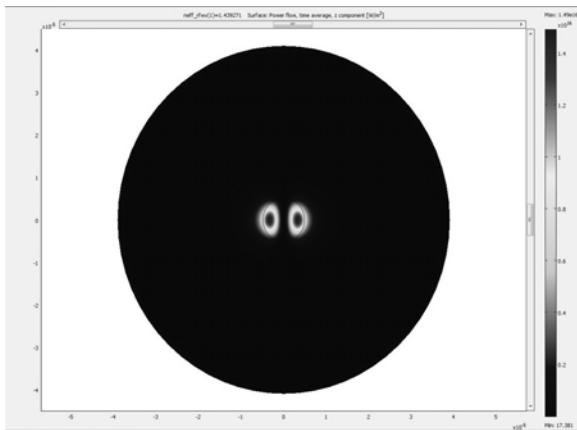


Fig. 20 The numerical simulated time averaged laser power flow across the transverse section of a single mode optical fiber with a core of  $8.82\ \mu\text{m}$  and  $11.33\ \mu\text{m}$  axes and a cladding of  $70.59\ \mu\text{m}$  and  $90.67\ \mu\text{m}$  axes

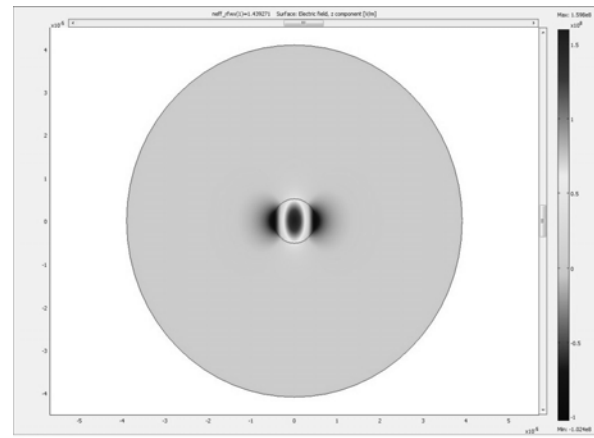


Fig. 21 The numerical simulated time averaged laser electric field distribution into the transverse section of a singlemode optical fiber with a core of  $8.82\ \mu\text{m}$  and  $11.33\ \mu\text{m}$  axes and a cladding of  $70.59\ \mu\text{m}$  and  $90.67\ \mu\text{m}$  axes

The procedure tried during numerical simulation consists in considering the laser beam propagation along the bending such as the optical fiber appears as of an elliptical cross section.

The deformation was considered by imposing a mechanical stress/pressure on the external surface of the plastic protection layer deposited on the glass cladding. The deformation is expressed in  $\mu\text{m}$ . The deformed dimensions of the glass cladding and core (the ellipse axes) are calculated as the area remains constant. The maximum value of the considered plastic layer deformation (denoted as strain) was of  $20\ \mu\text{m}$ .

## VII. CONCLUSION

The results of the DFB-FL sensor simulation proves that we obtained a realistic model of the sensor. The effects of the mechanical deformation (bending the optical fiber) were put in evidence. Fig. 18-21 reveal that important modifications in laser power flow and electric field distributions appear as effect of microdeformations applied to the studied optical fiber.

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